## **Milestone Review Flysheet**

## PDR, CDR, FRR

Institution Name	Massachusetts Institute of Technology

Milestone	DUB
Minestone	PDK

Vehicle Properties		
Diameter (in)	6.26	
Length (in)	124.5	
Gross Liftoff Weight (lb)	46.28	
Launch Lug/button Size	16-Mar	
Motor Retention	Stainless Steel Rod	

Motor Properties				
Motor Manufacturer	Cesaroni			
Motor Designation	L1115			
Max/Average Thrust (N/lb)	1713N, 1119N			
Total Impulse (N-sec/lb-sec)	5015 Ns			
Mass pre/post Burn (lb)	4.4kg, 2.01kg			

Stability Analysis			
Center of Pressure (in from nose)	89.95		
Center of Gravity (in from nose)	77.09		
Static Stability Margin	2.11		
Thrust-to-Weight Ratio	8.3		
Rail Size (in) / Length (in)	1.5" x 1.5", 8ft		

Ascent Analysis		
Rail Exit Velocity (ft/s)	54	
Max Velocity (ft/s)	594.6	
Max Mach Number	0.53	
Max Acceleration (ft/s^2)	503.3	
Peak Altitude (ft)	5,650	

Recovery System Properties					
	Dr	ogue Parach	ute		
Manufactu	ırer/Model	Surplus Military Pilot			
Size			60"		
Altituo	le at Deployn	nent (ft)	5,650		
Velocity at Deployment (ft/s)			~20 ft/s depending on wind		
Terminal Velocity (ft/s)			48.4		
Recovery Harness Material			Tubuluar Nylon		
Harness Size/Thickness (in)			1"		
Recove	ry Harness Le	ength (ft)	16'		
Harness/Airframe of paylo Interfaces attached		of payload sa	Attached to a 3/8" forged eye bolt in top of payload sabot unit. Eye bolt also attached to webbing that goes to tender descender.		
Kinetic Energy During Descent (ft-lb)	Section 1	Section 2	Section 3	Section 4	
	1331.3				

Recovery System Properties				
Main Parachute				
Manufactu	ırer/Model		RocketMan	
Si	ze		R16 Classic	
Altitud	de at Deploym	nent (ft)	3000'	
Velocit	y at Deploym	ent (ft/s)	48.4	
Lan	ding Velocity	(ft/s)	13.9	
Recov	ery Harness M	<b>M</b> aterial	Tubular Nylon	
Harne	ss Size/Thickr	ness (in)	1"	
Recove	ry Harness Le	ngth (ft)	3.25'	
	Harness/Airframe Interfaces  38" Forged Eye nut into top of avionics by connected to threaded rod into top of motors and the second sec			-
Kinetic Energy Upon Landing	Section 1	Section 2	Section 3	Section 4
(ft-lb)	70.7 (body)	.8 (Nose/Sab	69.7 (Quadrotor)	

Recovery System Properties		
Electronics/Ejection		
Altimeter(s) Make/Model	Featherweight Raven2	
Redundancy Plan	Perfectflite Stratologger, redundant drogue ejection charges. Redundant igniters in Tender Descender on main	
Pad Stay Time (Launch Configuration)	Upwards of 8 hours	

Recovery System Properties				
Electronics/Ejection				
Rocket Locators (Make, Model)	Big Red Bee 70cm Transmitter/2m GPS.			
Transmitting Frequencies	***Required by CDR***			
Black Power Mass	3.5			
Drogue Parachute (gram)				
Black Power Mass	0.2 (Tender Descender release device)			
Main Parachute (gram)				

## **Milestone Review Flysheet**

PDR, CDR, FRR

Institution Name	Massa	chusetts Institute of Technology	Milestone	PDR	
		Payload/S	cience		
Succinct Overview of Pay Experiment	A Custom built composite quadrotor carrying a system of payloads to exhibit object tracking and recogning the composite of Payload/Science in dynamic environments and to quantitatively measure high-altitude atmospheric lightning events				
The primary payload of the rocket will be the quadrotor which will be equipped with GPS, radio telecommand devices, an IMU, pressure sensors, an onboard computes, CMOS cameras, IR emitter detectors, and an autonomous flight stabilizer.					
Mass of Payload/Sc	The mass of the quadrotor is being estimated to be about 10 pounds.  Mass of Payload/Science				
		Test Plan Sche	dula/Status		
		Early December (Likely December 8)			
Ejection Charge Te	est(s)	Early December (Energy December 6)			
Sub-scale Test Flights		18-Nov			
Early February Full-scale Test Flights					
		Additional C	omments		